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THE HYDROFOIL CORPORATION

89 A WEST STREET

ANNAPOLIS, MD.

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TECHNICAL NEMORANDUM No. HM-19

Gilruth Boat

Prepared for: Office of Naval Research Washington, D. C. Contract No. Nonr-13601

November 20, 1951

Copy No. 10

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November 20, 1951 CONFIDENTIAL

THE HYDROFOIL CORPORATION

TECHNICAL MEMORANDUM HM-19

Subject: Gilruth Boat

Prepared for

Office of Naval Research

Washington, D. C.

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Frederic E. Bolliger

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THE HYDROFOIL CORPORATION

TECHNICAL MEMORANDUM HM-19

November 20, 1951

Subject:

Gilruth Boat

Prepared for:

Office of Naval Research, Washington, D. C.

Under Contract No. Nonr-13601

By:

Frederic E. Bolliger

References:

HFC Drawing No. 1067

1. Introduction.

A hydrofoil craft designed as a sail boat and known as the Gilruth Boat became the property of The Hydrofoil Corporation. It was decided to convert the boat to an ergine driven, experimental hydrofoil craft.

2. Design Features.

a. Platform.

For simple and inexpensive manufacture combined with low weight and good rigidity a wooden structure with plywood reinforcements was used for the platform which was mounted on the catamarans, struts and foils of the boat.

The pilot's seat is adjustable fore and aft on the platform.

b. Propulsion.

The craft is equipped with a standard 5 H.P. outboard motor which is carried by a coil spring balanced parallelogram that permits ready depth adjustment of the propeller.

A float chamber is built around the engine to guard against submergence of the engine.

c. Controls.

The controls are arranged as follows:

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The right hand of the pilot operates:

- 1. Main stick to control elevation flap on rear foil.
- 2. Rotatable knob on main stick to control engine throttle and spark advance.

The left hand operates:

- 3. Auxiliary stick to control dual rudder.
- 4. Rope and cam action jam cleat to control engine position.

The feet operate:

5. Dual pedals to control the ailerons.

3. Physical Data.

	7 24 0#
Length overall	13'2"
Beam (Span of main foil)	11'0"
Draft with crew, maximum	2'10"
Weight of craft	405 lbs.
Weight of crew	180 lbs.
Weight of craft with crew	585 lbs.
Main foil NACA 64-208	
Chord, main foil	12.5"
Thickness, main foil	1.0"
Area, projected, main foil	10.9 sq. ft.
Rear foil with elevation flap NACA 64-208	
Chord, rear foil	9.25m
Thickness, rear foil	0.7hm
Area, projected, rear foil	3.95 sq. ft.
Ailerons, NACA 647-412	
Chord, ailerons	6.0 ⁿ
Thickness, ailerons	0.72
Area, projected, ailerons (each)	1.25 sq. ft.
Estimated speed on foils	5 - 7 kmots

4. Load Distribution and Foil Loading.

a. With pilot in seat and seat in full aft position:

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68 lbs.

17.2 lbs/sq.ft.

The main foil and ailerons carry 86.8% of the weight 508 lbs.

The main foil loading is 37.9 lbs/sq.ft.

The rear foil carries 13.2% of the weight 77 lbs.

The rear foil loading is 19.5 lbs/sq.ft.

b. With pilot in seat and seat in full forward position:

The Main foil and ailerons carry 88.3% of the weight 517 lbs.

The main foil loading is 38.6 lbs.

5. Propulsion Unit.

The propulsion unit consists of a standard Mercury KF-5 outboard motor with a rated output of 5 H.P. at 4200 R.P.M.

The propeller has a 6 3/4^m diameter and 6 1/2^m pitch.

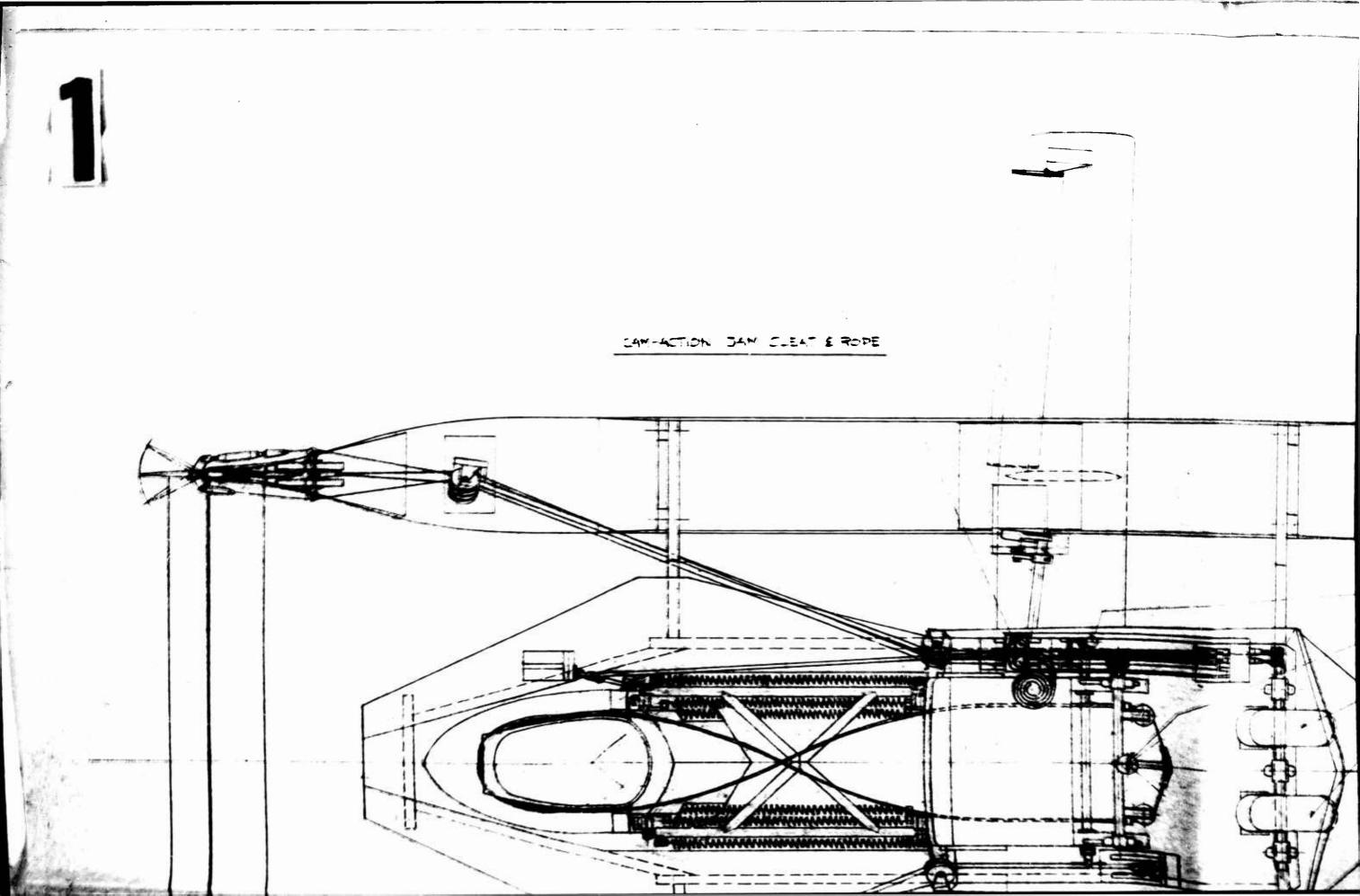
The rear foil carries 11.7% of the weight

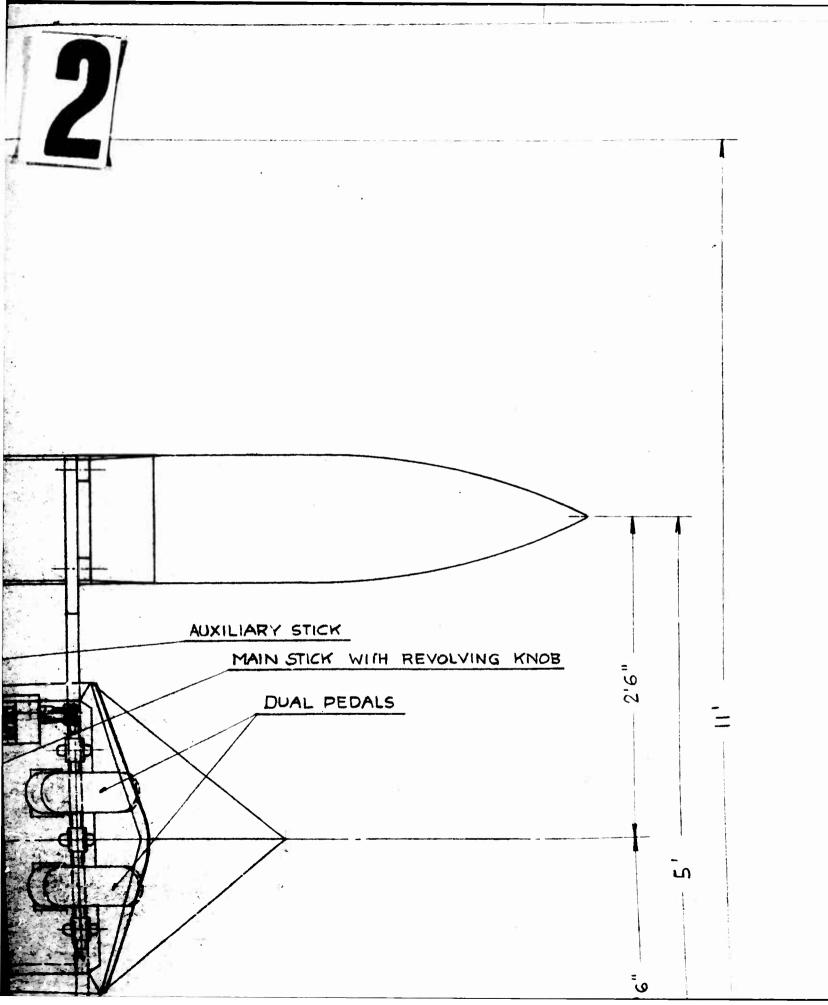
The rear foil loading is

6. Conclusion.

The craft has been tested under varying conditions. The craft can be flown easily with manual controls. The performance of the craft in straight flight and in turns is very satisfactory.

Frederic E. Bolliger





BEAM (SPAN OF MAIN FOIL)
MAX. DRAFT WITH CREW (A

WEIGHT OF CRAFT WEIGHT OF PILOT WEIGHT OF CRAFT WITH

MAIN FOIL NACA 64-20 CHORD THICKNESS AREA PROJECTED

REAR FOIL NACA 64-2 CHORD THICKNESS AREA

AILERONS NACA 64,-4
CHORD
THICKNESS
AREA PROJECTED (EA

LOAD DISTRIBUTION WIT SEAT FULL AFT MAIN FOIL CARRIES

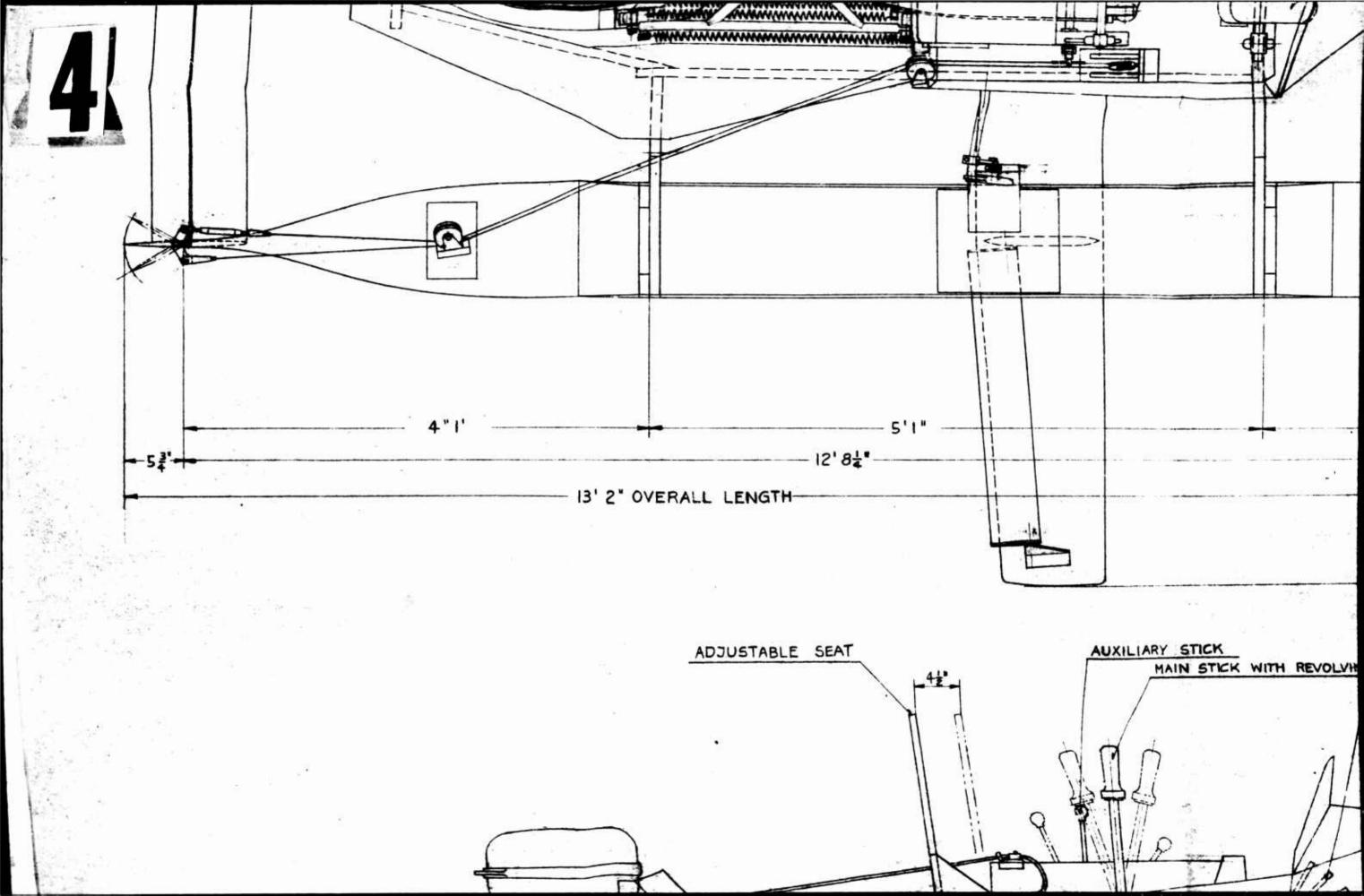
REAR FOIL CARRIES
SEAT FULL FORE (41/2)
MAIN FOIL CARRIES
REAR FOIL CARRIES

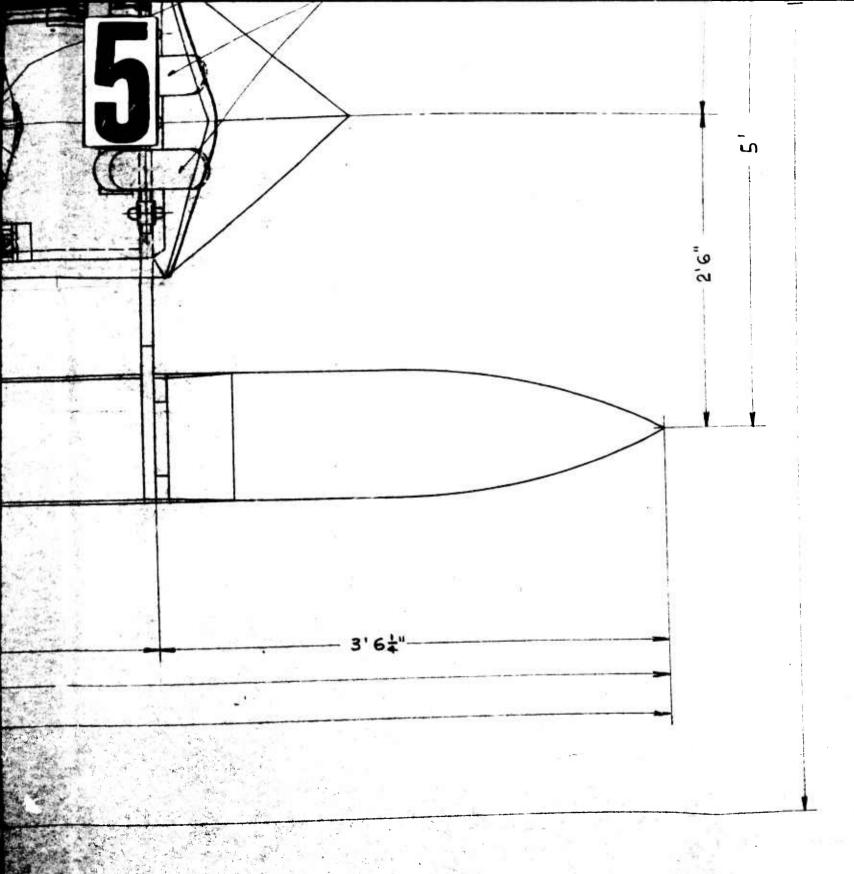
PROPULSION

LENGTH OVERALL BEAM (SPAN OF MAIN FOIL) MAX. DRAFT WITH CREW (AFLOAT)	13' 2" 11' 2'10"	
WEIGHT OF CRAFT WEIGHT OF CRAFT WITH CREW	405 180 585	LBS.
MAIN FOIL NACA 64-208 CHORD THICKNESS AREA PROJECTED	12.5 " 1 " 10.9	SQ.FT.
REAR FOIL NACA 64-208 CHORD THICKNESS AREA	9,25 " .74" 3.95	SQ.FT.
Allerons NACA 64,-412 CHORD THICKNESS AREA PROJECTED (EACH)	6 " .72" 1.25	SQ.FT.
LOAD DISTRIBUTION WITH CREW SEAT FULL AFT MAIN FOIL CARRIES 508 LBS REAR FOIL CARRIES 77 LBS SEAT FULL FORE (4½*)	86.8 13.2	% %
MAIN FOIL CARRIES 517 LBS REAR FOIL CARRIES 68 LBS	88.3 11.7	% %

161 Fg.

.





STICK WITH REVOLVING KNOB DUAL PEDALS

THICKNESS - AREA PROJECTE

LOAD DISTRIBUTION SEAT FULL AFT MAIN FOIL CA REAR FOIL CA SEAT FULL FOR MAIN FOIL REAR FOIL

PROPULSION MERCURY KF 5 PROPELLER D

MOTOR SUPPORT SPRING BALANC MAX. VERTICAL

CONTROLS :

RIGHT HAND OF A. STICK FOR

B. REVOLVIN

AND SPAR

LEFT HAND OF

C. AUXILIAR

D. ROPE FO

FEET OPERATE

E. DUAL PE

SEAT FULL FORE (41)

MAIN FOIL CARRIES 517 LBS REAR FOIL CARRIES 68 LBS

38.3 11,7

PROPULSION

MERCURY KF5 OUTBOARD MOTOR PROPELLER DIA x PITCH

H.P. $6\frac{3}{4} \times 6\frac{1}{2}$ IN.

MOTOR SUPPORT!

SPRING BALANCED PARALLELOGRAM MAX. VERTICAL MOVEMENT

2'

COMTROLS :

RIGHT HAND OPERATES:

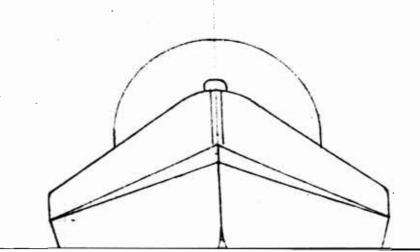
- A. STICK FOR ELEVATION FLAP CONTROL
- B. REVOLVING KNOB FOR ENGINE THROTTLE AND SPARK ADVANCE CONTROL

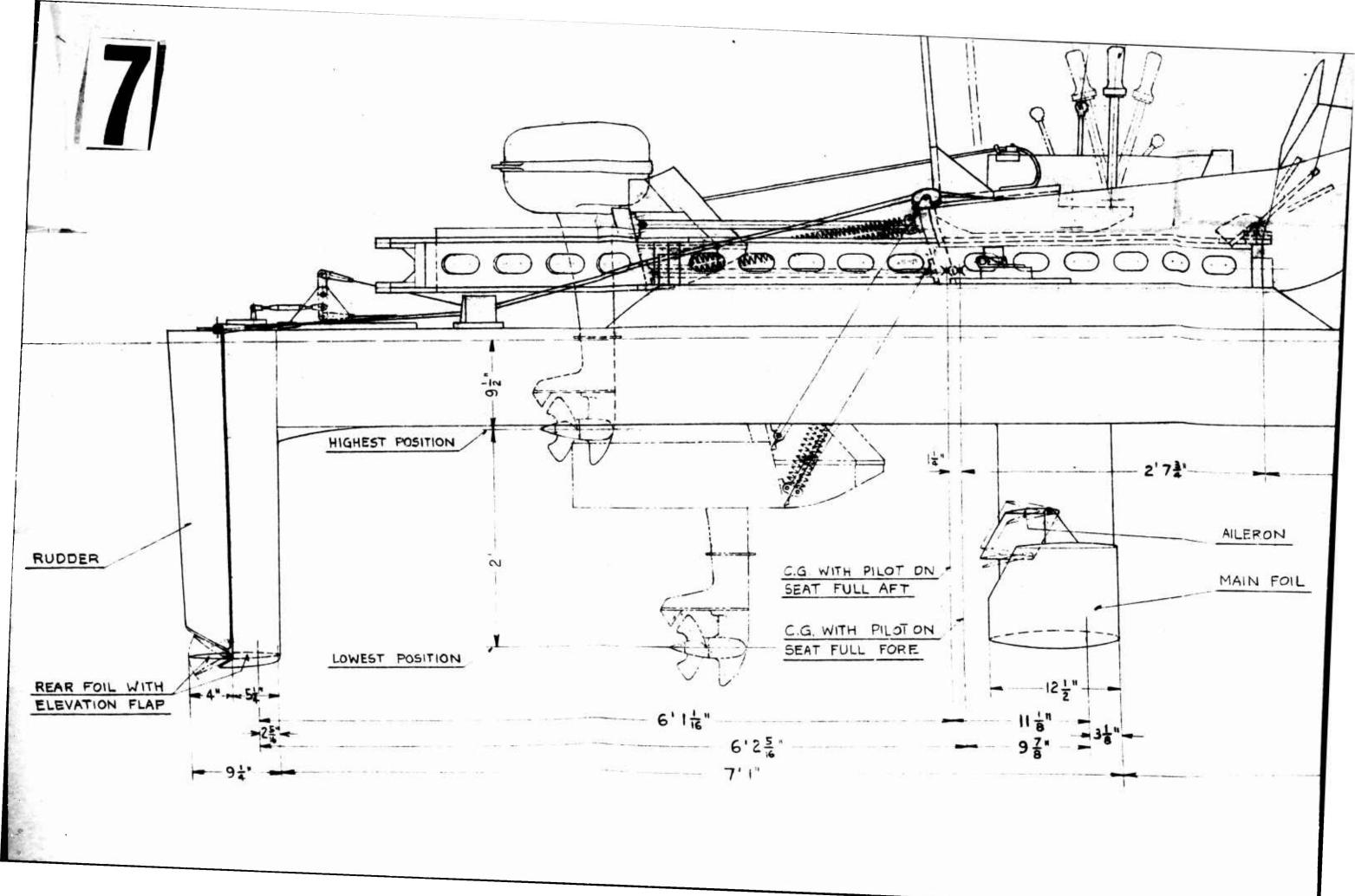
LEFT HAND OPERATES :

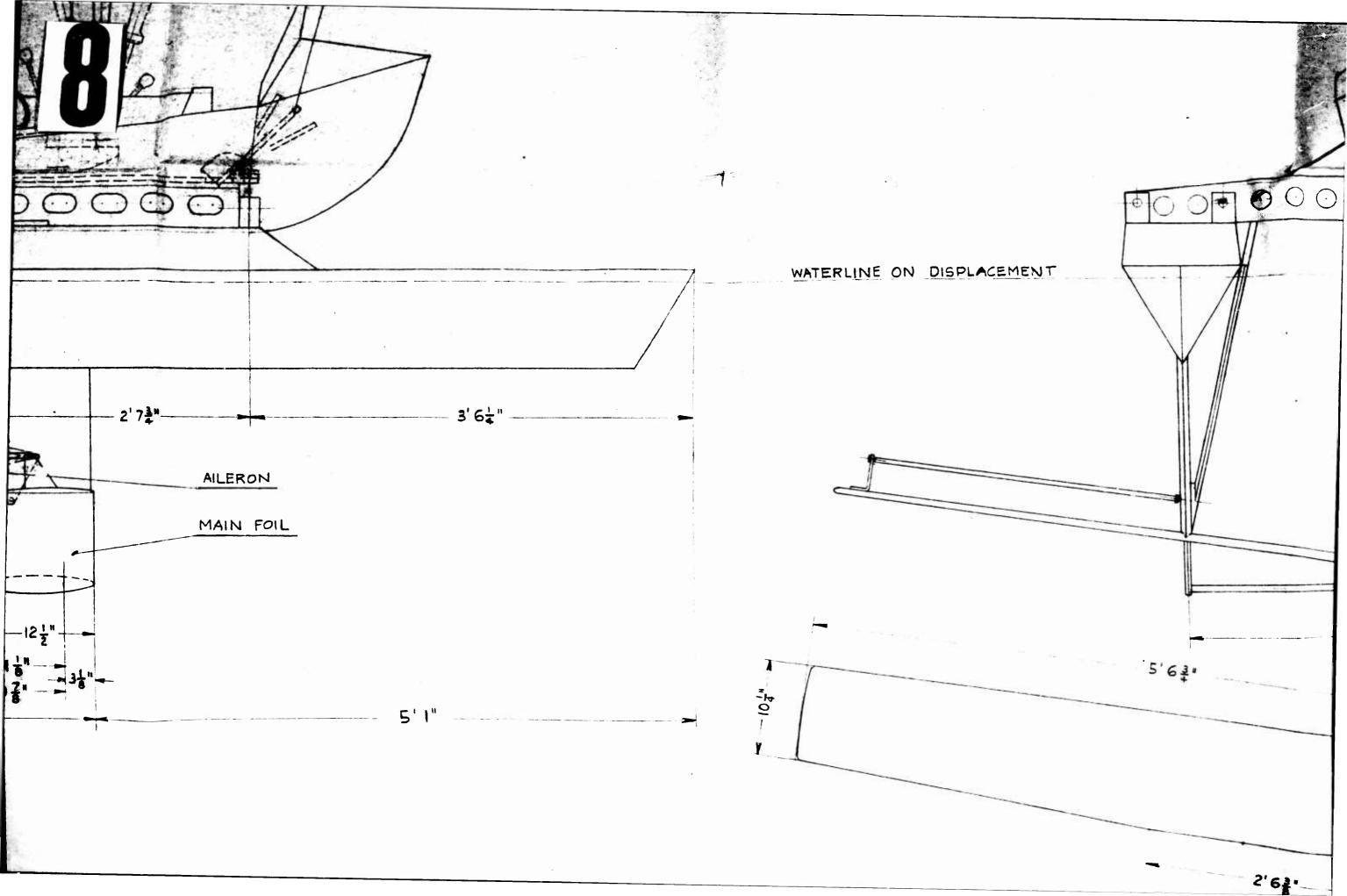
- C. AUXILIARY STICK FOR DUAL RUDDER CONTROL
- D. ROPE FOR ENGINE LOWERING.

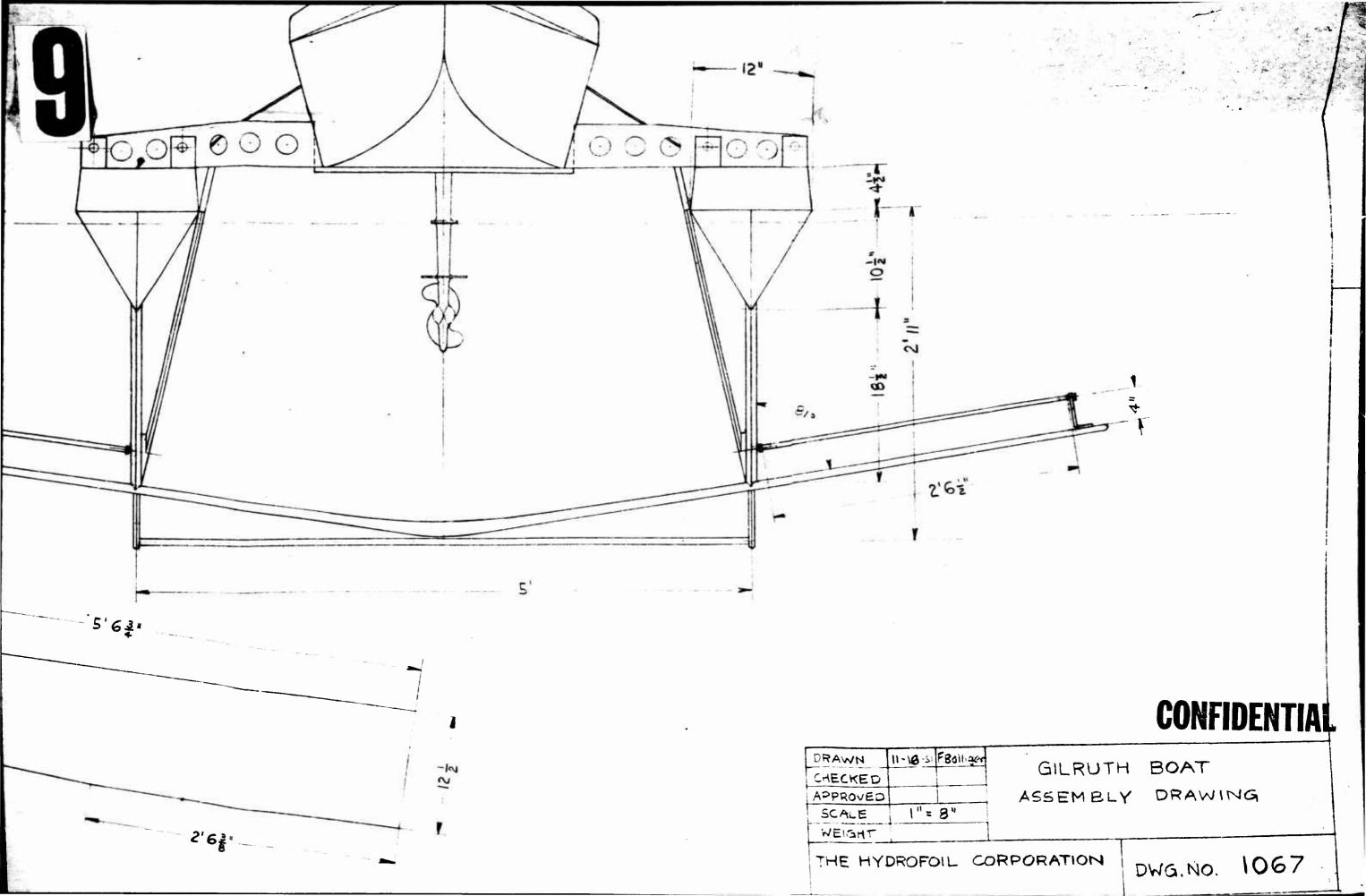
FEET OPERATE:

E. DUAL PEDAL FOR AILERONS CONTROL









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